Aerospace Structures and Design Lab Autoorelazione(s1() Conceptual Design of Aircraft —stA-sY. 2019/2020



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Fire Fighting Drone wants to be an innovative idea in the area of anti-fire. The flagship features of this drone are: low cost, fast, Path

Initial design

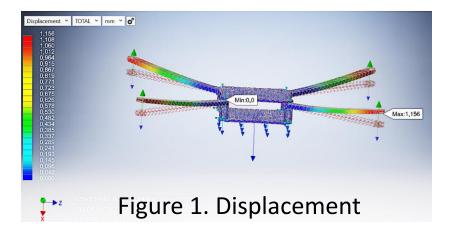
- The configuration of the drone is a **quadcopter**
- The model was made with the software Autodesk Inventor
 Professional 2018
- The material chosen for the frame is a carbon fiber epoxy
- The initial weight estimation of the model (except the payload) is about 6 kg.

Structural Analysis

It was made a structural Analysis with the software Inventor Nastran 2018, in *take off* condition

Mission profile

- The drone take off from an elevated basement until he reaches a height of 30 m;
- A cruise phase follows to get to the fire.
- Now the drone drop the fireballs, one by one, one the fire;
- After that he will go back to the basement



Configuration

It was compared the viable configurations
 with a full factorial design of experiment and
 has been

chosen the configuration with the highest value of:

 $Kd = \frac{N^{\circ}fb \ Range}{Price^2}$

	Config N°	Batteria	motore	N° fireballs	Prezzo \$	Range
1		B1	M1	7	1308	10.8Km
2		B2	M2	7	2360	22Km
4	*	B2	M1	10	2006	14.3Km
3		B1	M2	5	1392	7Km

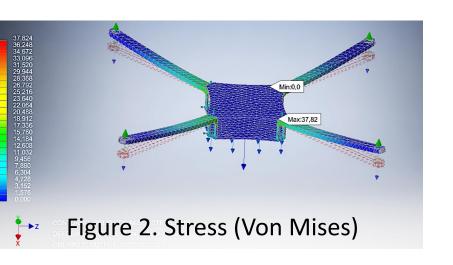
Table 1. Possible configurations

Propeller

- An appropriate choice of the size of the propeller (R
 - = 48 cm) guarantees maximum range and avoid
 - critical velocity in correspondence of the tip of the



This analysis has provided the results in terms of **critical stress** (37.8 MPa) and **maximum displacement** (1.2 mm).

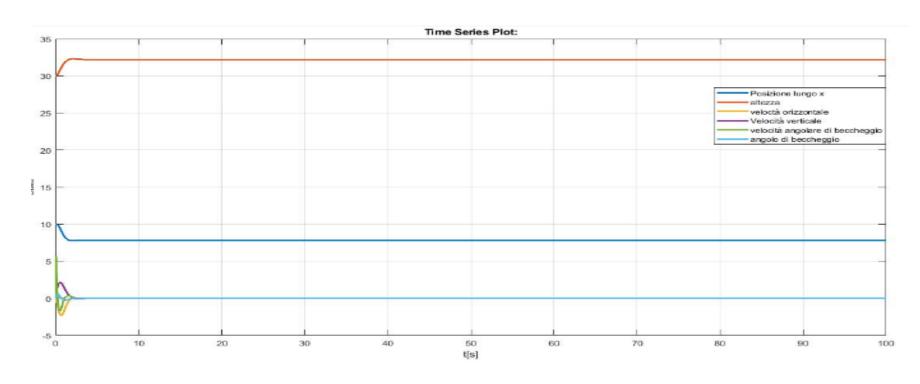


Payload

- The *fireball* is able to extinguish a fire of 3 m³ (or an area of 1,5 m²)
- Each fireball weigh 1.3 kg
- The drone must be able to carry a payload of about 10 kg (about 7 fireballs).

Flight dynamics control (LQR control)

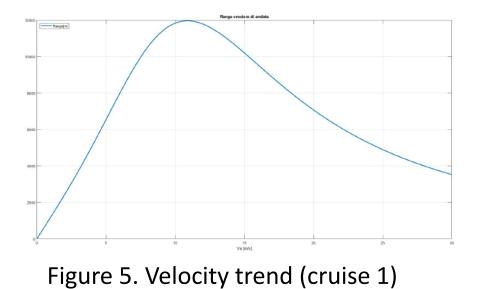
Dynamics Longitudinal response to a gust (considering the gust as an implulse signal)

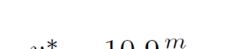


blade.

Operating Velocities

- The operating velocities guarantee the **maximum range** during the two cruise phases (with and without payload)





$$v_1^* = 10.9 \frac{m}{s}$$

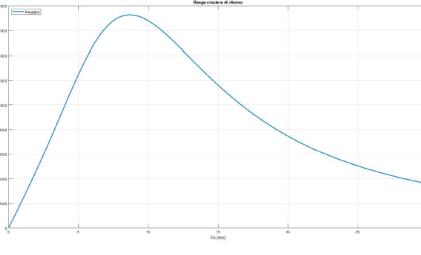


Figure 6. Velocity trend (cruise 1)

$$v_2^* = 8.6 \frac{m}{s}$$

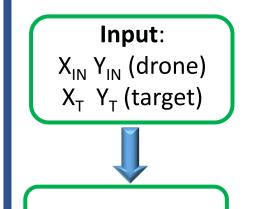
Performance

Motore	Batteria	$\operatorname{Range}_{MAX}$	Prezzo	N°_{fb}
T motor P60 Kv170	Tattus Plus	10.8 Km	1308\$	7

Table 2. Best configuration

Path planning

- **Objective:** To minimize the trajectory of the drone with the aim of reducing cost.
- **Domain:** It is a bidimensional (x,y) map of an area of the city, within which there are static obstacle (ex. palace) that the drone must avoid.



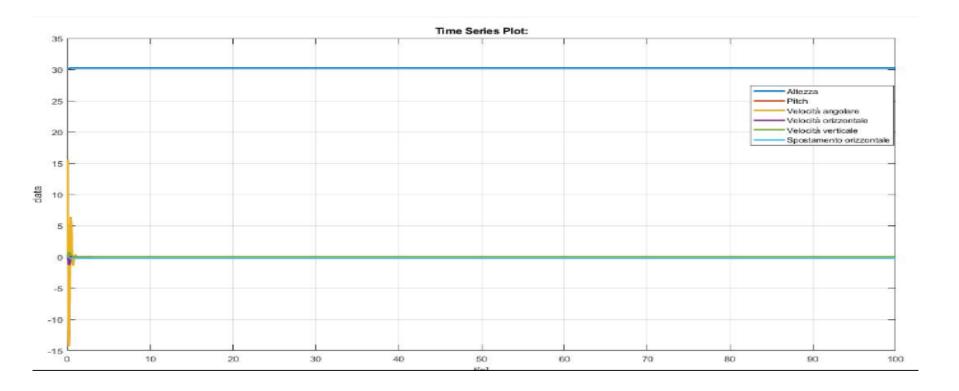
Definitions in genetic algorithm:

- **Chromosome:** It correspond to a trajectory from the initial point (drone's basement) to the target point (the fire);
- **Gene:** The 2D map is discretized in a finite number of reference points that the drone can reach;

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Figure 4. Propeller

- ✓ The stability of the drone has been achieved. The responses of the system converge close to the desired values.
- Dynamics Longitudinal response to small variations in weight. It has been considered the variation of the weight after releasing a single fireball.



✓ The stability of the drone has been achieved. The responses of the system converge to the desired values.

